3.1 INTRODUCTION

3.1.1 GENERAL

This Chapter contains checklists as well as the description of recommended procedures to be followed in the event of an emergency. Engine failure or other airplane-related emergencies are most unlikely to occur if the prescribed procedures for pre-flight checks and airplane maintenance are followed.

If, nonetheless, an emergency does arise, the guidelines given here should be followed and applied in order to clear the problem.

As it is impossible to foresee all kinds of emergencies and cover them in this Airplane Flight Manual, a thorough understanding of the airplane by the pilot is, in addition to his knowledge and experience, an essential factor in the solution of any problems which may arise.

WARNING

In each emergency, control over the flight attitude and the preparation of a possible emergency landing have priority over attempts to solve the current problem ("first fly the aircraft"). Prior to the flight the pilot must consider the suitability of the terrain for an emergency landing for each phase of the flight. For a safe flight the pilot must constantly keep a safe minimum flight altitude. Solutions for various adverse scenarios should be thought over in advance. Thus it should be guaranteed that the pilot is at no time shocked by an engine failure and that he can act calmly and with determination.



3.1.2 CERTAIN AIRSPEEDS IN EMERGENCIES

	Event	Flight Mass	850 kg 1874 lb	1000 kg 2205 lb	1150 kg 2535 lb	1200 kg 2646 lb
I	Engine failure after take-off (Flaps T/O)		59 KIAS	66 KIAS	72 KIAS	74 KIAS
I	Airspeed for best glide angle (Flaps UP)		60 KIAS	68 KIAS	73 KIAS	76 KIAS
	Emergency	Flaps UP	60 KIAS	68 KIAS	73 KIAS	76 KIAS
I	landing with	Flaps T/O	59 KIAS	66 KIAS	72 KIAS	74 KIAS
I	engine off	Flaps LDG	58 KIAS	63 KIAS	71 KIAS	73 KIAS

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3.2 ENGINE PROBLEMS

3.2.1 ENGINE PROBLEMS ON THE GROUND

1.	Throttle	IDLE
2.	Brakes	as required
3.	Engine	switch off, if considered
		necessary; otherwise
		establish the cause of the
		problem and re-establish
		engine performance

CAUTION

If the oil pressure is below the green sector, the engine must be switched off immediately.

WARNING

If the problem cannot be cleared, the airplane must not be flown.

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3.2.2 ENGINE PROBLEMS DURING TAKE-OFF

(a) Take-Off Can Still Be Aborted (Sufficient Runway Length Available)					
Land Straight Ahead:					
1. Throttle IDLE					
On the Ground:					
2. Brakes as required					

CAUTION

If sufficient time is remaining, the risk of fire in the event of a collision can be reduced as follows:

-	Fuel tank selector	OFF
-	Mixture control lever	LEAN - shut engine off
-	Ignition switch	OFF
-	Master switch (ALT/BAT)	OFF

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(b) Take-Off Can No Longer Be Aborted

1.	Airspeed	 	 	74 KIAS (1200 kg, 2646 lb)
				72 KIAS (1150 kg, 2535 lb)
				66 KIAS (1000 kg, 2205 lb)
				59 KIAS (850 kg, 1874 lb)

WARNING

If, in the event of an engine problem occurring during take-off, the take-off can no longer be aborted and a safe height has not been reached, then a straight-ahead emergency landing should be carried out. Turning back can be fatal.

If Time Allows:

2.	Fuel tank selector	check selected tank
3.	Electrical fuel pump	check ON
4.	Ignition switch	check BOTH
5.	Throttle	check MAX PWR
6.	RPM lever	check HIGH RPM
7.	Mixture control lever	check RICH (leaner above
		5000 ft)
8.	Alternate Air	OPEN

WARNING

If the problem does not clear itself immediately, and the engine is no longer producing sufficient power, then an emergency landing must be carried out.

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3.2.3 ENGINE PROBLEMS IN FLIGHT

(a) Engine Running Roughly

WARNING

An engine which is running very roughly can lead to the loss of the propeller. If the engine is running roughly operation should only be continued if there is no other alternative.

I	1.	Airspeed	76 KIAS (1200 kg, 2646 lb) 73 KIAS (1150 kg, 2535 lb) 68 KIAS (1000 kg, 2205 lb) 60 KIAS (850 kg, 1874 lb)
	 3. 4. 6. 8. 9. 	Electrical fuel pump Fuel tank selector Engine instruments Throttle RPM lever Mixture control lever Alternate Air Ignition status light	check selected tank check check check set for smooth running OPEN

10. Ignition switch check BOTH

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11. Ignition circuit breaker (IGN) pull (only if the electronic ignition control unit is installed); if rough running is cleared by doing this, the circuit breaker should remain open

12. Throttle/RPM/Mixture try various settings

WARNING

If the problem does not clear itself immediately, and the engine is no longer producing sufficient power, then an emergency landing should be carried out.

(b) Loss of Oil Pressure

- 1. Check oil pressure warning light and oil pressure indicator.
- 2. Check oil temperature.
 - 2a. If the oil pressure indication drops below the green sector and the oil temperature is normal (oil pressure warning light does not illuminate or flash):
 - * Monitor the oil pressure warning light: it is probable that the oil pressure indication is defective.
 - * Monitor the oil and cylinder head temperatures.
 - 2b. If the oil pressure indication drops below the green sector while the oil or cylinder head temperature is rising, or

If the oil pressure warning light illuminates or flashes, or

If both of these occur together:

- * Reduce engine power to the minimum required.
- * Land as soon as possible.
- * Be prepared for engine failure and emergency landing.
- 2c. Oil pressure tending to zero combined with:

Vibration, loss of oil, possibly unusual metallic noise and smoke:

- * A mechanical failure in the engine is apparent.
- * Shut off engine immediately and
- * Carry out emergency landing in accordance with 3.5.1 EMERGENCY LANDING WITH ENGINE OFF.

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(c) High Oil Pressure

Check oil temperature.

* If the oil temperature is normal, it is probable that the fault lies in the oil pressure indication, which should thus be ignored (the airplane should be serviced).

END OF CHECKLIST

(d) High Oil Temperature

Check cylinder head and exhaust gas temperature.

- * If neither of these is high, it is probable that the fault lies in the oil temperature indication. The airplane should be serviced. A stable oil temperature indication of 26 °F (-3 °C) or 317 °F (158 °C) suggests a failure of the oil temperature sensor.
- * If the cylinder head temperature or exhaust gas temperature is also high:
 - Check oil pressure. If the oil pressure is low, proceed as in 3.2.3 (b) -LOSS OF OIL PRESSURE.
 - If the oil pressure is in the green sector:
 - Check mixture setting, enrich mixture if necessary.
 - Reduce power; if this produces no improvement, land at the nearest appropriate airfield.

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(e) High Cylinder Head Temperature

Cylinder head temperature in yellow sector or above:

- 1. Check mixture setting, enrich mixture if necessary.
- 2. Check oil temperature.
 - * If the oil temperature is also high:
 - Check oil pressure. If the oil pressure is low, proceed as in 3.2.3 (b) -LOSS OF OIL PRESSURE.
 - If the oil pressure is in the green sector:
 - Reduce power; if this produces no improvement, land at the nearest appropriate airfield.
 - Be prepared for possible emergency landing.



(f) High RPM

RPM moves on its own into the yellow sector, or is in the red sector:

- 1. Check friction adjuster for throttle quadrant.
- 2. Check oil pressure: Following a loss of oil or oil pressure, the propeller governor sets a high RPM. In this case the RPM should be regulated using the throttle. Proceed as in 3.2.3 (b) LOSS OF OIL PRESSURE.
- 3. If oil pressure is normal:
 - * Pull RPM lever back and listen for an associated drop in RPM:
 - If the indication does not change in spite of an audible drop in RPM, it is probable that the RPM indication is defective, which should thus be ignored (the airplane should be serviced).
 - If there is no audible drop in RPM, it is probable that the governor system is defective. In this case the RPM should be regulated using the throttle.

Emergency

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(g) Loss of RPM

1.	Electrical fuel pump	check ON
2.	Fuel tank selector	check
3.	Friction adjuster for throttle quadrant	check sufficiently tight
4.	RPM lever	HIGH RPM

- Listen for rise in RPM.
 - If there is no audible rise in RPM, it is probable that the governor system is defective. In this case the RPM can be regulated within certain limits using the throttle.
 - Land at the nearest appropriate airfield.
 - Be prepared for possible emergency landing.
 - If the indication does not change in spite of an audible rise in RPM, it is probable that the RPM indication is defective, which should thus be ignored (the airplane should be serviced).



(h) High Fuel Flow

Fuel flow in the red sector:

- - * If the fuel pressure is low, refer to 3.2.3 (i) LOW FUEL PRESSURE WITH THE ELECTRICAL FUEL PUMP SET TO ON.
 - * If the fuel pressure is in the green sector, or the fuel pressure warning light is not illuminated, the likely cause is a defective fuel flow indication, which should thus be ignored (the airplane should be serviced). Fuel flow data should be taken from the engine performance table in Chapter 5.
- 3. Check fuel quantity. A rapid reduction in fuel quantity confirms a high fuel flow.



(i) Low Fuel Pressure with the Electrical Fuel Pump Set to ON

Fuel pressure warning light illuminates, or fuel pressure	indication below the green sector:
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- 1. Fuel flow check:
 - * If the fuel flow is high, there is possibly a leak (between the injection system and the injectors). Land on the nearest suitable airfield.
 - * If the fuel flow is in the green sector and the engine is running smoothly, the likely cause is a defective fuel pressure indication, which should thus be ignored (the airplane should be serviced).

Monitor engine for power loss and rough operation that could indicate fuel starvation. If the engine is no longer producing sufficient power, then an emergency landing should be carried out.

3.2.4 RESTARTING THE ENGINE WITH WINDMILLING PROPELLER

NOTE

Restarting the engine is possible at all airspeeds above 70 KIAS up to v_{NE} and up to the maximum demonstrated operating altitude.

NOTE

As long as an airspeed of at least 65 KIAS is maintained, and there is no major engine failure, the propeller will continue to windmill.

1.	Airspeed	80 KIAS
2.	Fuel tank selector	fullest tank
3.	Ignition switch	check BOTH
4.	Mixture control lever	check appropriate position
5.	Electrical fuel pump	check ON
6.	Alternate air	OPEN

If Engine Does Not Start:

7	Mixture control lever	LEAN
/ .	wixiure conitoriever	 IFAIN

8. Mixture control lever push forward slowly until

engine starts

NOTE

If it is not possible to start the engine:

- Adopt glide configuration as in 3.4 GLIDING.
- Carry out emergency landing as in 3.5.1 EMERGENCY LANDING WITH ENGINE OFF.

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3.2.5 DEFECTIVE ENGINE CONTROLS

Defective Mixture Control Cable

- (a) Flight and Landing:
 - 1. Maintain altitude to the nearest airfield.
 - 2. During descent, test the reaction of the engine to a higher power setting. A lean mixture can lead to engine roughness and a loss of power. The landing approach must be planned accordingly.

WARNING

Go-around may become impossible with the remaining power.

(b) Engine Shut-Down:

1.	Parking brake	set
2.	Engine instruments	check
3.	Avionics master switch	OFF
4.	All electrical equipment	OFF
5.	Throttle	IDLE
6.	Ignition switch	OFF
7.	Master switch (ALT/BAT)	OFF

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Defective Throttle Control Cable

- (a) Sufficient Engine Power Available to Continue Flight:
 - 1. Approach nearest airfield, control engine power with RPM lever.
 - 2. Perform landing with shut-down engine.
- (b) No Sufficient Engine Power Available to Continue Flight:
 - 1. Carry out emergency landing as in 3.5.1 EMERGENCY LANDING WITH ENGINE OFF.

END OF CHECKLIST

<u>Defective RPM Lever Control Cable</u>

- (a) Sufficient Engine Power Available to Continue Flight:
 - 1. Approach nearest airfield, control engine power with throttle.
 - 2. Perform normal landing.

WARNING

Go-around may become impossible with the remaining power.

- (b) No Sufficient Engine Power Available to Continue Flight:
 - Carry out emergency landing as in 3.5.1 EMERGENCY LANDING WITH ENGINE OFF.

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3.2.6 RESTARTING THE ENGINE WITH STATIONARY PROPELLER

NOTE

Restarting the engine is possible at all airspeeds above 80 KIAS up to v_{NE} and up to the maximum demonstrated operating altitude.

1.	Airspeed	80 KIAS
2.	Electrical equipment	OFF
3.	Avionics master switch	OFF
4.	Master switch (BAT)	check ON
5.	Mixture control lever	check
6.	Fuel tank selector	check
7.	Electrical fuel pump	check ON
8.	Alternate air	OPEN
9.	Ignition switch	START

NOTE

By increasing the airspeed above approximately 130 KIAS, the propeller will begin to rotate and the engine can thus be started. For this, the Ignition switch should be set at BOTH (see 3.2.4 - RESTARTING THE ENGINE WITH WINDMILLING PROPELLER). An altitude loss of at least 1000 ft (300 meter) must be allowed for.

if it is not possible to start the engine:

- Adopt glide configuration as in 3.4 GLIDING
- Carry out emergency landing as in 3.5.1 EMERGENCY LANDING WITH ENGINE OFF.

CONTINUED

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CAUTION

Engine restart following an engine fire should only be attempted if it is unlikely that a safe emergency landing can be made. It must be expected that engine restart is impossible after an engine fire.

Procedures

3.3 SMOKE AND FIRE

3.3.1 SMOKE AND FIRE ON THE GROUND

(a) Engine Fire When Starting on the Ground

1.	Fuel tank selector	 OFF
2.	Cabin heat	 OFF
2	Brakes	annly

3. Brakes apply

After Standstill:

4.	Throttle	MAX PWR
----	----------	---------

5. Master switch (ALT/BAT) OFF

When the Engine Has Stopped:

6.	Ignition switch	O	FF
7.	Canopy	0	oen

8. Airplane evacuate immediately



(b) Electrical Fire with Smoke on the Ground

1.	Master switch (ALT/BAT)	OFF
If the Eng	ine is Running:	
2.	Throttle	IDLE
3.	Mixture control lever	LEAN - shut off engine
When the	Engine Has Stopped:	
4.	Ignition switch	OFF
5.	Canopy	open
6.	Airplane	evacuate immediately



3.3.2 SMOKE AND FIRE DURING TAKE-OFF

(a) If Take-Off Can Still Be Aborted

1. Throttle IDLE

2. Cabin heat OFF

3. Brakes apply - bring the airplane to

a stop

4. After stopping proceed as in 3.3.1 -

SMOKE AND FIRE ON THE

GROUND

END OF CHECKLIST

(b) If Take-Off Cannot Be Aborted

- 1. Cabin heat OFF
- 2. If possible, fly along a short-cut traffic circuit and land on the airfield.

WARNING

If, in the event of an engine problem occurring during take-off, the take-off can no longer be aborted and a safe height has not been reached, then a straight-ahead emergency landing should be carried out. Turning back can be fatal.

72 KIAS (1150 kg, 2235 lb)

66 KIAS (1000 kg, 2205 lb)

59 KIAS (850 kg, 1874 lb)

CONTINUED

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After Climbing to a Height From Which the Selected Landing Area Can Be Reached Safely:

4.	Fuel tank selector	OFF
5.	Electrical fuel pump	OFF
6.	Cabin heat	OFF
7.	Master switch (ALT/BAT)	OFF
8.	Emergency window(s)	open if required
9.	Carry out emergency landing with engine off.	Allow for increased landing
	distance due to the flap position.	

CAUTION

In case of extreme smoke development, the front canopy may be unlatched during flight. This allows it to partially open, in order to improve ventilation. The canopy will remain open in this position. Flight characteristics will not be affected significantly.



3.3.3 SMOKE AND FIRE IN FLIGHT

CAUTION

In the event of smoke or fire, prepare to land the airplane without delay while completing fire suppression and/or smoke evacuation procedures. If it cannot be visually verified that the fire has been completely extinguished, whether the smoke has cleared or not, land immediately at the nearest suitable airfield or landing site.

(a) Engine Fire in Flight

- 1. Cabin heat OFF
- 2. Select appropriate emergency landing field.

When it Seems Certain that the Landing Field Will Be Reached:

- 3. Fuel tank selector OFF
- 4. Throttle MAX PWR
- 5. Electrical fuel pump OFF
- 6. Master switch (ALT/BAT) ON
- 7. Emergency window(s) open if required
- 8. Carry out emergency landing with engine off.

CAUTION

In case of extreme smoke development, the front canopy may be unlatched during flight. This allows it to partially open, in order to improve ventilation. The canopy will remain open in this position. Flight characteristics will not be affected significantly.

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(b) Electrical Fire with Smoke in Flight

1.	Emergency switch	ON if installed
2.	Master switch (ALT/BAT)	OFF
3.	Cabin heat	OFF
4.	Emergency window(s)	open if required
5.	Land at an appropriate airfield as soon as pos	ssible.

CAUTION

Switching OFF the Master switch (ALT/BAT) will lead to total failure of all electronic and electric equipment. Also affected from this are - if installed - the attitude gyro (artificial horizon) and the directional gyro.

However, by switching the Emergency switch ON (only installed in the IFR model), the emergency battery will supply power to the attitude gyro (artificial horizon) and the flood light.

CAUTION

In case of extreme smoke development, the front canopy may be unlatched during flight. This allows it to partially open, in order to improve ventilation. The canopy will remain open in this position. Flight characteristics will not be affected significantly.

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3.4 GLIDING

1.	Flaps	UP
2.	Airspeed	76 KIAS (1200 kg, 2646 lb)
		73 KIAS (1150 kg, 2535 lb)
		68 KIAS (1000 kg, 2205 lb)
		60 KIAS (850 kg, 1874 lb)

NOTE

The glide ratio is 8.8; i.e., for every 1000 ft (305 meter) of altitude loss the maximum horizontal distance traveled in still air is 1.45 NM (2.68 km). During this the propeller will continue to windmill.

With a stationary propeller the glide ratio is 10.3; this corresponds to a maximum horizontal distance of 1.70 NM (3.14 km) for every 1000 ft altitude. In consideration of a safe airspeed however, this configuration may not be attainable.

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3.5 EMERGENCY LANDINGS

3.5.1 EMERGENCY LANDING WITH ENGINE OFF

I			CAUTION	
l I			gency landing the adjustable back ixed in the upright position.	rests (if installed)
] 1]]	1.	Adjustable b	packrests (if installed)	adjust to the upright position described by a placard on the roll-over bar and verify proper fixation.
	2.	Select suital	ole landing area. If no level landir	ng area is available, a landing
		on an upwai	d slope should be sought.	
3	3.	Consider wi	nd.	
	1.	Approach:	If possible, fly along a short-cut if downwind leg of the circuit the latinspected for obstacles from a strong offset at each part of the circuit is direction to be assessed.	anding area should be uitable height. The degree of
. 5	5.	Airspeed .		76 KIAS (1200 kg, 2646 lb) 73 KIAS (1150 kg, 2535 lb) 68 KIAS (1000 kg, 2205 lb) 60 KIAS (850 kg, 1874 lb)
1 6	3.	If time allow	s	advise ATC
1 7	7.	Fuel tank se	lector	OFF

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When It Is Certain That the Landing Field Will Be Reached:

8.	Flaps	LDG
9.	Safety harnesses	tighten

CAUTION

If sufficient time is remaining, the risk of fire in the event of a collision with obstacles can be reduced as follows:

- Ignition switch OFF Master switch (ALT/BAT) OFF
- 10. Touchdown with the lowest possible airspeed

3.5.2 LANDING WITH A DEFECTIVE TIRE ON THE MAIN LANDING GEAR

CAUTION

A defective (e.g. burst) tire is not usually easy to detect. The damage normally occurs during take-off or landing, and is hardly noticeable during fast taxiing. It is only during the roll-out after landing or at lower taxiing speeds that a tendency to swerve occurs. Rapid and determined action is then required.

- 1. Advise ATC.
- Land the airplane at the edge of the runway that is located on the side of the intact tire, so that changes in direction which must be expected during roll-out due to the braking action of the defective tire can be corrected on the runway.
- 3. Land with one wing low. The wing on the side of the intact tire should be held low.
- 4. Direction should be maintained using the rudder. This should be supported by use of the brake. It is possible that the brake must be applied strongly if necessary to the point where the wheel locks. The wide track of the landing gear will prevent the airplane from tipping over a wide speed range. There is no pronounced tendency to tip even when skidding.

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3.5.3 LANDING WITH DEFECTIVE BRAKES

In general, a landing on grass is recommended in order to reduce the landing run by virtue of the greater rolling resistance.

CAUTION

If sufficient time is remaining, the risk of fire in the event of a collision can be reduced as follows:

- Fuel tank selector OFF

- Mixture control lever LEAN - shut off engine

- Ignition switch OFF

- Master switch (ALT/BAT) OFF

3.6 RECOVERY FROM AN UNINTENTIONAL SPIN

CAUTION

Steps 1 to 4 must be carried out **immediately** and **simultaneously**.

	1.	Throttle	IDLE	
	2.	Rudder	full deflection against	
			direction of spin	
	3.	Elevator (control stick)	fully forward	
	4.	Ailerons	neutral	
	5.	Flaps	UP	
When Rotation Has Stopped:				
	6.	Rudder	neutral	
	7.	Elevator (control stick)	pull carefully	
	8.	Return the airplane from a descending into a	normal flight attitude. In so	
		doing do not exceed the 'never exceed speed	l', v _{NE} .	

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3.7 OTHER EMERGENCIES

3.7.1 **ICING**

<u>Unintentional Flight Into Icing Conditions</u>

١.	Leave the iding area (by changing aithdue of turning back, in order to	
	reach zones with a higher ambient temperatu	re).
2.	Pitot heating	ON
3.	Cabin heat	ON
4.	Air distributor lever	▲ (up)
5.	RPM	increase, in order to prevent
		ice build-up on the propeller
		blades
6.	Alternate air	OPEN

1 Leave the icing area (by changing altitude or turning back, in order to

CAUTION

7. Emergency window(s) open if required

Ice build-up increases the stalling speed. If required for safety reasons, engine speeds up to 2700 RPM are admissible without time limit.

8. ATC advise if an emergency is expected

CAUTION

When the Pitot heating fails, and the alternate static valve is installed:

- Alternate static valve	OPEN
- Emergency window(s)	close

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3.7.2 FAILURES IN THE ELECTRICAL SYSTEM

(a) Complete Failure of the Electrical System

Due to the strong mechanical design as well as due to the required check of the system during scheduled inspections, a total failure of the electrical system is extremely unlikely. If, nevertheless, a total failure should occur, all circuit breakers should be checked, pulled and re-set. If this does not help:

- Set Emergency switch to ON (if installed).
- When necessary, use the flood light for lighting the instruments as well as levers and switches, etc.
- Set power based on lever positions and engine noise.
- Prepare landing with flaps in the given position.
- Land on the nearest appropriate airfield.



(b) Alternator Failure

An alternator failure is indicated by an illuminated or flashing alternator warning light (ALT or ALTERNATOR) on the annunciator panel and a flashing ammeter on the Vision Microsystems VM 1000 engine instrument.

(i) Alternator Failure During Flight

1.	Circuit breakers	check; if all are OK, proceed
		with step 2
2.	Electrical equipment	switch OFF all equipment
		which is not needed
3.	Voltmeter	check regularly

CAUTION

Those items of equipment which are not needed for the safe operation and secure landing of the airplane can be switched off with the Essential Bus switch (if installed). When the essential bus is switched ON, only the following items of equipment are supplied with power:

- NAV/COM 1.
- Transponder (XPDR).
- Flood light.
- Attitude gyro (artificial horizon).
- VM 1000 engine instrument.
- Annunciator panel.
- GPS (if installed).
- Landing light.
- Pitot heating system.
- Flaps.

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These items of equipment can be supplied with power by the battery for at least 30 minutes. Economical use, in particular of the Pitot heating, and switching off equipment that is not needed extends the time during which the other equipment remains available. During the 30 minutes period, the airplane must be landed at a suitable airfield.

For cases in which the battery capacity is not sufficient to reach a suitable airfield, an emergency battery is installed in the IFR model, serving as an additional back-up system for the attitude gyro (artificial horizon) and flood light. This battery is switched on with the Emergency switch. It lasts for 1 hour and 30 minutes when the flood light is switched on.

(ii) Alternator Failure on the Ground

NOTE

An alternator failure may also be indicted on ground with the engine running on IDLE.

1. Engine speed 1200 RPM

2. Electrical equipment OFF

3. Ammeter check

If the caution light does not extinguish, and the ammeter flashes and reads zero:

- Terminate flight preparation.



(c) Starter Malfunction

If the starter does not disengage from the engine after starting (starter warning light (START) on the annunciator panel remains illuminated or flashing after the engine has started):

1.	Throttle	IDLE
2.	Mixture control lever	LEAN - shut off engine
3.	Ignition switch	OFF
4.	Master switch (ALT/BAT)	OFF

Terminate flight preparation!

END OF CHECKLIST

(d) Overvoltage

If a voltage in the upper red sector (above 32 Volts) is indicated:

1.	Essential bus	ON, if installed
2.	Master switch (ALT)	OFF

WARNING

Leave Master switch (BAT) ON!

- Equipment that is not needed,in particular Pitot heating OFF
- 4. Land on the nearest appropriate airfield.

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3.7.3 SUSPICION OF CARBON MONOXIDE CONTAMINATION IN THE CABIN

Carbon monoxide (CO) is a gas which is developed during the combustion process. It is poisonous and without smell. Since it occurs however usually together with fuel gases, it can be detected. Increased concentration of carbon monoxide in closed spaces can be fatal. The occurrence of CO in the cabin is possible only due to a defect. If a smell similar to exhaust gases is noticed in the cabin, the measures in the checklist below should be taken:

The DA 40 may be equipped with a CO detector (optional equipment, OÄM 40-253). If the visual alert annunciator illuminates in flight, press the TEST/RESET button. If the alert continues with the remote light staying ON or a smell similar to exhaust gases is noticed in the cabin, the following measures should be taken:

1.	Cabin heat	OFF
2.	Ventilation	open
3.	Emergency window(s)	open
4.	Forward canopy	open

CAUTION

In case of suspicion of carbon monoxide contamination in the cabin, the front canopy may be unlatched during flight. This allows it to partially open, in order to improve ventilation. The canopy will remain open in this position. Flight characteristics will not be affected significantly.

NOTE

The presence of carbon monoxide is indicated by a visual alarm if OÄM 40-253 is carried out.

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3.7.4 'DOOR'-WARNING LIGHT ON

I	1. 2. 3.	Airspeed reduce immediately Canopy
<u>Cano</u>	oy L	<u>Jnlocked</u>
 	4. 5.	Airspeed below 140 KIAS Land at the next suitable airfield.
<u>Rear</u>	Doo	r Unlocked
 	4. 5.	Airspeed below 140 KIAS Land at the next suitable airfield.
i		WARNING
 		Do not try to lock the rear door in flight. The safety latch may disengage and the door opens. Usually this results in a separation of the door from the airplane.
I		NOTE
 		If the rear door has been lost the airplane can be safely flown to the next suitable airfield.

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3.7.5 EMERGENCY EXIT

In case of a roll-over of the airplane on ground, it can be evacuated through the rear door. For this purpose release the front hinge of the rear door. The function is displayed on a placard next to the hinge.